

IMPACTS OF HIGH PRESSURE GASES IN CATALYSIS PROCESS

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ABSTRACT

This paper is dedicated to produce information and brilliant trial information of catalytic combustion at operational gas turbine conditions. The underlying assignment of the paper is to plan and develop a high weight burning test facility, where the catalytic combustion trials can be performed at genuine gas turbine conditions. Considering this, a very propelled burning test facility has been composed, developed and tested. This test facility is fit for recreating ignition conditions applicable to an extensive variety of working gas turbine conditions and various types of fuel gasses. The state of the combustor (test area) is like a "can" sort gas turbine combustor, however with huge contrasts in its kind of operation. The test combustor is relied upon to work at close adiabatic ignition conditions and there will be no augmentations of cooling, weakening or optional supply of air into the burning procedure. The geometry of the combustor comprises of three fundamental zones, for example, air/fuel blending zone, synergist response zone and downstream gas stage response zone with no distinction of the mass stream at channel and exit. Fuel adaptability of the applications was additionally thought about in the outline stage and legitimate measures have been taken so as to use two sorts of focused energizes, methane and gasified biomass.

I INTRODUCTION

The main interest of this study is on the development of environmental friendly, fuel flexible and economical combustion systems for land based gas turbine applications. With that in mind, this work concentrates on the advancement of synergist burning innovation for gas turbine ignition. The fundamental work depicted here is the exploratory examinations of the operation of burning impetuses under gas turbine working conditions. Synergist ignition is generally considered as the burning innovation for up and coming age of gas turbines. The inspiration for said improvements is its capacity of ultra low level of contamination emanations. Aside from the outflow issue, synergist burning evade a considerable lot of the specialized issues related with gas turbine ignition, for example, fire acoustics, fire solidness issues, mechanical vibrations and so forth., sensibly and combustor structures are made rather straightforward as well. The innovation additionally offers specific quality for usage of non-regular turbine powers of low warming limit [Johansson et al. 1998].

A very propelled burning test facility has been outlined and developed for reenacting states of present and future gas turbines for the exploratory examinations of reactant ignition. Fuel adaptability of the applications was considered in

the outline stage, bookkeeping from regular powers to inexhaustible powers; petroleum gas and gasified biomass, where a noteworthy contrast of the warming an incentive from one to alternate must be considered. Burning examinations are directed at weights as high as 35 bars in the fuel limit of 100 kW. The test facility configuration has additionally been considered for the conceivable outcomes of incorporating reactant burning trials into the future propelled gas turbine cycle states of evaporative gas turbine (EvGT) or humidified gas turbine (HAT) [Dalili 2003]. The ignition impetuses subjected for the high weight burning analyses were produced and given by the undertaking accomplice, the Division of Chemical Technology, KTH. The exploratory outcomes revealed in this work are primarily examinations on reactant burning of methane on very dynamic valuable metal under genuine gas turbine conditions. Reactant burning of gasified biomass was additionally tried under pressurized conditions. A portion of the underlying outcomes are incorporated herewith.

Gas Turbine Power in Energy Industry

Gas turbines are one of the world's most imperative wellsprings of mechanical and local power, they are likewise the predominant main thrust behind flight. The interest for gas turbine units from the vitality business is developing relentlessly. As indicated by the announced figures, gas turbine unit deals have been expanded by 37% amid the years 1999 and 2000. By 1999, the offer of gas turbine and consolidated cycle control plants (GT/CCPP) in control age were still of unobtrusive extents as same as atomic power station and just about half as much as the introduced hydro limit.

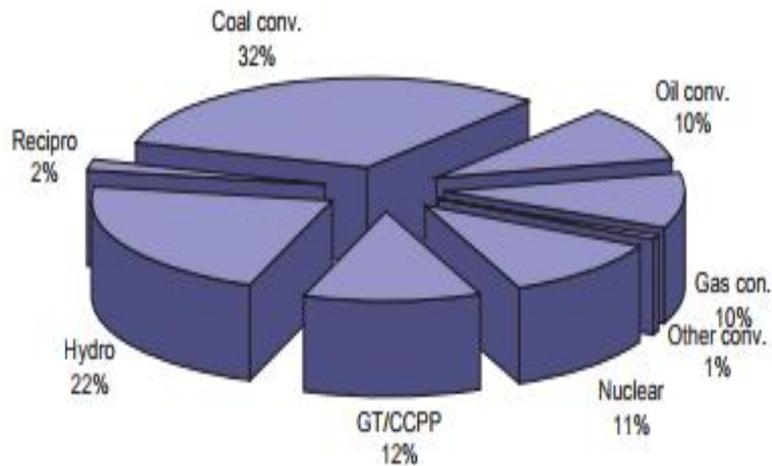


Figure 1: Installed capacity and fuel resources used

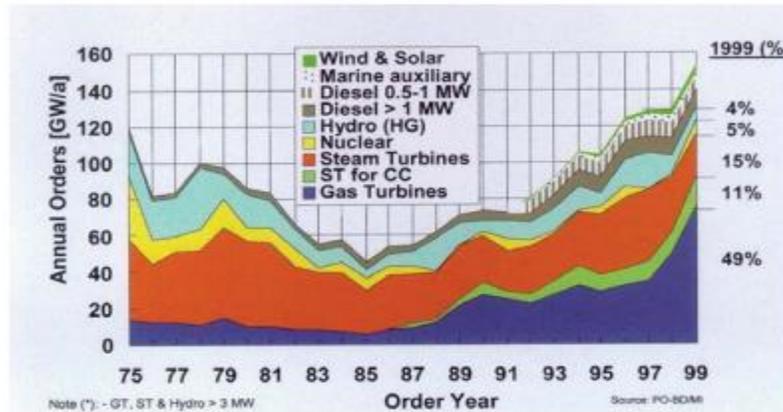


Figure 2: Additional capacity per year by technology

II BACKGROUND

Gas turbine is a versatile machine in energy and transport sectors, being considered as the prime mover of 21st century [Lefebvre, 1998]. History of the gas turbines in power generation runs not more than 70 years back, even though it was born more than a century ago. Norwegian/Swedish citizen Jens William Ægidius Elling is credited for producing the world's first gas turbine which produced net positive power in 1903 [CompEduHPT, 2004]. Figure 1 illustrates the schematic of Ellings gas design and figure 2 shows the prototype of Ellings gas turbine design from 1903 and 1912 are now being exhibited at Norsk Teknisk Museum in Oslo [Wikipedia, 2013]. Many years later, in 1930, Sir Frank Whittle, building on the early work of Elling, managed to build a design for practical gas turbine engine for an airplane, the jet engine.

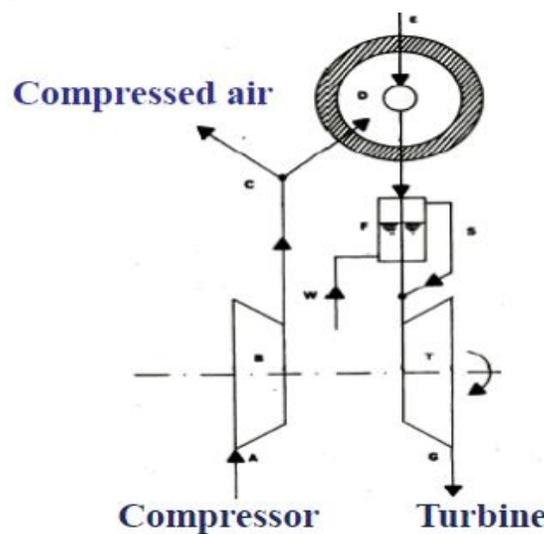


Figure 3: Schematic view of the World's first gas turbine (Elling's Gas Turbine)

III ENVIRONMENTAL IMPACTS OF GAS TURBINE COMBUSTION

The environmental effects caused by the emanations of gas turbine establishments are of awesome concern. Having fuel burning as the primary procedure of vitality change, gas turbines are normally anticipated that would emanate outflows. Emanations of the gas turbines are normally oxides of nitrogen (NO and NO₂ on the whole called as NO_x), carbon monoxide (CO), un-consumed hydrocarbon (UHC), carbon dioxide (CO₂) and water vapor (H₂O). Gaseous petrol is the most well-known and broadly utilized fuel for control age gas turbines and it is for the most part considered as a perfect fuel. However the nature of common gasses changes relying upon the areas they are separated and now and again it contains a few measures of nitrogen mixes and sulfur. Nearness of such mixes in the fuel is dangerous in the feeling of outflow control. Gas turbines likewise use different sorts of powers, fluid fills, for example, diesel or vaporous energizes like gasified coal or gasified biomass. In these circumstances discharges are higher in contrast with the circumstances where petroleum gas is utilized for the burning. Besides, these different sorts of fills create extra discharges, for example, ash (a type of un-copied carbon) and sulfur oxide (SO₂). Oxides of nitrogen are today considered as the most hazardous toxin created by the burning procedure and it could differently influence the human/creature life and condition in a wide range of ways. Photochemical brown haze thus called corrosive downpours is immediate effects of NO_x outflows. Arrangements of the oxides of nitrogen are associated with the ignition procedure itself.

Carbon monoxide and un-consumed hydrocarbon discharges are consequences of fragmented ignition responses of the burning procedure. Outflows of these substances decrease the ignition proficiency. However these outflows can be decreased by giving adequate temperature and living arrangement time for the ignition procedure. Carbon monoxide is a notable poisonous gas. Outflows of hydrocarbon will connect with nitrogen oxide and daylight along these lines shaping ground level ozone. Outflows of hydrocarbons, for example, methane additionally bear high a dangerous atmospheric deviation potential. Sulfur dioxide is an immediate consequence of the nearness of sulfur in the fuel. Impact of the presence of sulfur dioxide emanations in the air can be portrayed by the arrangements of corrosive downpours. The main conceivable answer for take out the development of sulfur dioxide in the burning is to expel sulfur from the fuel itself. Arrangement of the principle outflow items; Carbon dioxide and water vapor can't be maintained a strategic distance from. Inferable from it being an ozone harming substance, emanation of carbon dioxide is a truly concerned issue with regards to a dangerous atmospheric deviation. However expanded process productivity would prompt diminished carbon dioxide development. This will give less measure of carbon dioxide per unit of warm vitality delivered.

IV REACTIVITY OF CATALYTIC SURFACES

The one of a kind imaging capacities of the STM are shown through the assurance of the exact area of the dynamic locales in three distinct procedures: Hydrodesulphurization on MoS₂, CO separation on Ni(111), and ethylene

separation on Ni(111). It is besides demonstrated how essential understanding into the nuclear subtle elements of reactant procedures can be used in the sane outline of better than ever impetuses. Cases of impetus configuration are given through surface alloying and also specific site blocking.

The reactivity of synergist surfaces is controlled by numerous parameters, for example, restricting energies of reactants, intermediates, and items, surface morphology, co-adsorbates, contaminations and so on. These distinctive impacts consolidate to give a tremendous parameter space for differing the reactivity of a surface. In surface science one for the most part tries to single out these parameters and utilize streamlined models to depict their impact on the reactivity. For instance the direct Brønsted-Evans-Polanyi connection between the initiation vitality and the response vitality showed by various rudimentary responses can be joined with straightforward dynamic models to give the reactivity as an element of the response vitality, which brings about alleged fountain of liquid magma plots. The impact of the surface morphology can be analyzed by concentrate the properties of various single-precious stone features, or as appeared in the accompanying area, even exceptionally neighborhood impacts identified with, e.g., single deformities on a surface can be examined. The mix of different point by point analyses can prompt a more profound comprehension of the systems representing the reactivity of a synergist surface. In any case, a definitive objective of surface science isn't just to have the capacity to portray how an impetus functions yet in addition to outline as good as ever synergist frameworks from first standards. In spite of the numerous examples of overcoming adversity of present day surface science, this objective is a long way from completely accomplished, despite the fact that a couple of illustrations do exist which could lead the path towards a levelheaded plan of impetuses. The last segment of this section manages cases of how this extreme errand inside heterogeneous catalysis might be drawn closer.

4.1 Reactant GAS TURBINES

Reactant fuel ignition innovation for gas turbine and warming applications was initially proposed by Pfefferle in mid seventies [Pfefferle, 1974, 1978]. Today, the innovation has achieved the level of business interests through many years of research and improvements. Inspiration for the improvement is its ultra low outflow capacity at bring down establishment and working expenses. A few research tasks and pilot plant operations of synergist burning were accounted for in various limits as completed by gas turbine produces together with impetus engineers. The world's first synergist combustor coordinated gas turbine has achieved showcase in 2002 and has been working since November 2002 [Catalytica Energy Systems, 2004. The (Xonon is the exchange name of the reactant framework created by Catalytica Energy Systems) prepared Kawasaki M1A-13X (1.4 MW) working at Sonoma Developmental Center is the primary business gas turbine in history to produce ultra-low emanations levels without the utilization of an extra fumes cleanup framework.

4.2 Synergist Combustor Developments

Despite the fact that the synergist burning innovation was proposed for gas turbine applications in mid 70s, it was not for all intents and purposes conceivable in gas turbine combustors at beginning periods. The early endeavors of creating reactant gas turbine combustors were obliged by necessity of high temperature synergist materials, that can remain on higher temperature levels exists in gas turbine conditions. Valuable metals, (Platinum, Palladium, Rhodium) surely understood synergist materials would not stand temperatures more than 800-900qC. Advancement of hexaaluminate materials by Japanese scientists presented in mid eighties [Machida et al., 1989] was a leap forward for consistent hunt on high temperature materials improvements. Hexaaluminate materials are accounted for to be fit for withstanding temperatures up to 1200qC without thermally falling apart its structure. The high temperature materials proposed for gas turbine applications are either hexaaluminate itself or hexaaluminates doped with metal oxides, which are called perovskites. In any case, there are no impetuses exhibit up to today for high temperature needs of present day high proficiency gas turbines, i.e. 1400 - 1500qC at the delta to the turbine area. As it is observed from writing, a few designing procedures have been presented on the gas turbine combustor advancements to defeat the material hindrances. Lean premixed ignition over the impetuses is one of the generally considered and spearheading methods coordinated to the gas turbine combustor improvements. The fuel is mostly changed over the impetuses, by constraining the warmth discharge and keeping away from over warming impetuses.

The nearness of a homogeneous burning zone at the downstream of the synergist bed apparently is a typical element of such outlines, where the gas stage responses increment the leave temperature as required by the turbine gulf conditions. A portion of the combustor models revealed in the writing are exhibited in the procedure segments of this section. Notwithstanding the high temperature soundness issue of the impetuses, another huge need is the high synergist action at low temperatures so as to light the air fuel blend at compressor leave temperatures, i.e. around 200 - 300qC at part stack conditions. Start temperature, normally canceled as light temperature of the impetus for the most part relies upon sort of fuel, blend quality and reactant materials. The condition of craftsmanship demonstrates the majority of the endeavors are to create synergist combustors for gas turbines running on flammable gas [Beebe et al., 1995], [Vortmeyer et al., 1996], [Yee et al., 2001]. Petroleum gas, mainstream fuel for arrive based power age applications, is thought to be a perfect fuel, which has methane as its fundamental constituent. Numerous specialists have revealed that the start of methane over the impetuses has appeared to be a troublesome errand at temperatures under 400qC even with exceedingly dynamic palladium based impetuses [Thevenin, 2002]. Reconciliation of preheating strategies to raise the air fuel blend temperature up to the impetuses light off point of confinement has turned into a fundamental requirement for gas turbine combustors intended for lean premixed ignition of gaseous petrol. The preheating methods used are either presentation of pre-burners or choice of recuperators [Fant et al., 2000]. Both of these systems have negative results and restrictions as the pre burner

operation adds to NO_x punishment while the recovery strategy is constrained to little scale turbines of weight proportions under 10 bars because of the issue of effectiveness drop with expanding weight proportions.

4.3 Reactant Combustor Designs

A few reactant combustor outlines have been proposed, created and tried in various synergist ignition/gas turbine joining ventures revealed in past decades. Those plans can fundamentally be cleared up into two noteworthy sorts, for example, completely synergist combustor outlines and cross breed combustor plans. An outline of those frameworks is found in figure 4. The completely synergist combustor changes over all the fuel over impetuses through heterogeneous responses and no fire related. Half and half combustor permits a fire at the downstream of the impetuses, where the total change of the fuel is occurred through homogeneous responses. The completely synergist combustor configuration was first proposed by Sadamori et. al [1995]. The proposed configuration was a co-operation between Osaka Gas, Kobe Steel and Catalysts and Chemicals Inc. [Johansson, 1998]. The significant imperative of the completely reactant configuration is the impediment on combustor leave temperature. The most extreme conceivable combustor outlet temperatures of around 1100°C are not satisfactory for the present need of high effectiveness gas turbines. Be that as it may, the completely reactant configuration is a promising procedure for ULE discharges in little scale turbines where the turbine bay temperatures (TIT) are not over 1100°C.

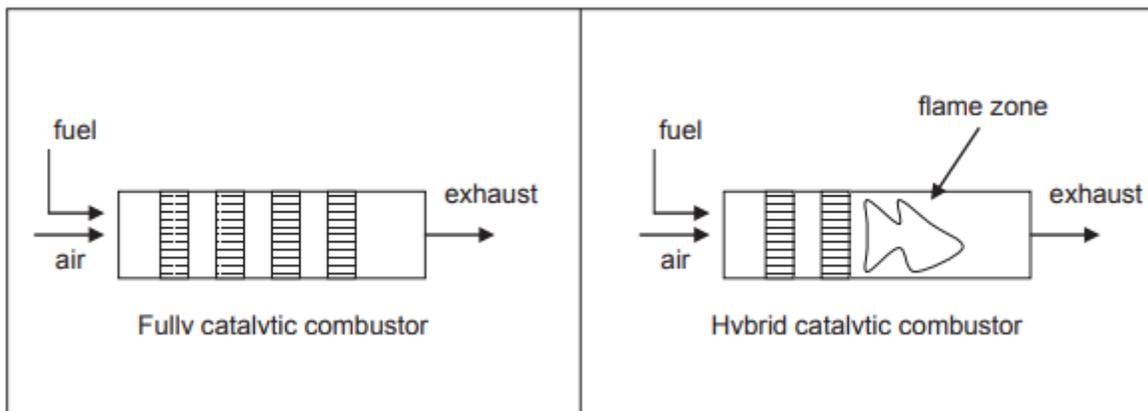


Figure 4: Basic designs of catalytic combustors

V CONCLUSIONS

An advanced combustion test facility has been designed, constructed and tested. The test facility is fit for reenacting ignition conditions important to an extensive variety of working gas turbine conditions and various types of gas fills. The combustor can be arranged to test the execution of individual impetus fragments and also full setups of reactant combustors, for example, completely synergist or half breed synergist combustor outlines. A PC based information

procurement framework is utilized for the precise estimations and to control temperature, weight, stream rates and discharges. One of the primary goals of this work was to tentatively explore the execution of ignition impetuses at conditions pertinent for gas turbine operations. By considering the operational necessities of the gas turbine combustors and in addition lining up with the known operational constraints of the impetuses, the test battle intended for examination of synergist burning exhibitions was partitioned into two classifications; testing of individual fragments of exceptionally dynamic impetuses for start reason and the multi sectioned designs of the impetus mixes to accomplish start and additionally the continuation of the concoction responses to fulfillment.

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